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PAUL LIEBERMAN

DoD *Top Secret* Clearance (IITRI)

DoD *Secret* Clearance (TRW)

DoD *Secret* Clearance (NTS)

Education

B.S. Aeronautical Engineering - New York University

M.S. Aeronautical Engineering - Princeton University

Ph.D. Applied Mechanics - Illinois Institute of Technology

Professional Experience

2000 -present EnisEnerGen LLC
1999 - present Lieberman Research Associates
1981 - 1999 NTS Engineering, Los Angeles, CA
1969 - 1981 TRW, Redondo Beach, CA
1957 - 1969 Illinois Institute of Technology/Research Institute (IITRI)
1955 - 1957 Experimental Rocket Engine Test Station, Edwards Air Force
 Base, CA (1st Lt USAF, Capt USAF Reserve)
1954 Summer Rocket Department, Bell Aircraft Corp, Buffalo, New York

Awards:

New York State Regents Scholarship (New York University)

Guggenheim Fellowship (Princeton University)

LRA Contracts

- Upgrade Performance Radiation Heat Transfer from Heatwux System to Road Asphalt, Greenwood Village, Colorado
- Hose-In-Hose Transfer Line Service Life DNFSB Response Support”, Defense Nuclear Facilities Safety Board, Washington River Protection Solutions, LLC (WRPS), Richland, Washington
- Liquid Nitrogen Test Facility Design for NASA Orbital Space Plane (Orbital Sciences Corp., Dulles, VA)
- Risk Management and Prevention Program Refrigeration Storage (Smithway Associates, Commerce, California)
- Buckling Test Design and Test Conduct of Orion Rocket Engine Columbiuim/Titanium Nozzle at High Temperature (Aerojet)
- Explosives Bunker Design (Dynamic Testing Inc (DTI) Site, Rustburg, VA) Liquid Nitrogen Blow-down Test Setup Design (NTS, Santa Clarita, CA))
- Plan and Conduct Lifetime Prediction Tests for Hose-in-Hose Transfer Lines (HIHTL) of
- Radioactive, Warm and Pressurized Waste Water (*CH2M-Hill*)

Specializations

- Compressed Air Energy Storage Systems
- Reduced Natural Gas Consumption for Combustion Turbines
- CO2 Sequestration
- Natural Gas Recovery from Gas Shale and Coal Beds
- Mine Clearance by Directed Strong Shock Pulses
- Toxic Cloud Capture and Incineration
- Liquid Hydrogen Testing

2. Method of transporting and storing wind generated energy using a pipeline

US 7,504,739 B2

March 17, 2009

US 7,755,212

July 13, 2010

(Pipeline)

The method involves storing and transporting power in the form of compressed air energy, via a pipeline. The method preferably consists of using at least one power source such as a wind turbine to drive a compressor to compress air into storage, wherein the size and length of the pipeline can be adapted to reduce the pressure losses that are experienced along the length of the pipeline. The facility or community using the energy can use energy in the form of electricity, or to drive pneumatic tools or equipment, or to generate chilled air as a by-product, which can be used for refrigeration, air conditioning or desalination. A utility or grid can be provided to generate compressed air energy when the wind is not blowing, wherein compressed air energy can be produced and stored during low demand periods, and used during high demand periods.

3. Method of Coordinating and Stabilizing the Delivery of Wind Generated Energy

US 6,963,802 B2

November 8, 2005

US 7,308,361

December 11, 2007

US 7,974,742

July 5, 2011

(Coordinating)

Gear Box in Wind Turbine nacelle drives generator, Use wind predictions, Emergency fuel burner, Recover compressor waste heat, Solar thermal collector, Mass of liquid inside pressure vessel, Output cold air to air compressor, Output air to refrigeration facility.

4. Method and Apparatus for Storing and Using Energy to Reduce the End-User Cost of Energy

US 7,155,912 B2

January 2, 2007

(Co-Generation)

Gear Box in Wind Turbine nacelle drives generator, Emergency fuel burner, Recover compressor waste heat, Solar thermal collector, Mass of liquid inside pressure vessel, Output cold air to air compressor, Output air to refrigeration facility, Time shifting the electrical power from grid.

5. Thermal energy storage system using compressed air energy and/or chilled water from desalination processes

US 7856843 B2

Dec 28, 2010

Australian 2014202086 (pending)

April 14, 2014

Chinese ZL200780046557.7

October 18, 2007

South Africa 2009/03446

October 18, 2009

(TES)

The invention relates to a universal system for producing cost effective energy particularly for cooling purposes. In one embodiment, wind turbines are used to generate electricity and compressed air energy, wherein the compressed air energy is used to co-generate electricity and chilled air. The chilled air is then used to chill water in either a mixing chamber, or a desalination system, wherein the chilled water is stored in a separation tank, wherein it can later be used to provide cooling for an air conditioning system for a facility. When desalination is used, the system produces chilled fresh drinking water which can be used for air conditioning, and then used as fresh drinking water. Any exhaust chilled air can be used directly for air conditioning.

6. Desalination method and system using compressed air energy systems

US 8,695,360 B2

April 15, 2014

US 8,863,547

October 21, 2014

Australian 2007238919 (pending)

March 30, 2008

South Africa 2008/09457

March 30, 2008

(Desalination)

The invention relates to a desalination method and system that uses freeze crystallization technology that incorporates the use of compressed air energy as the source for freezing temperatures. When compressed air is released by a turbo expander, chilled air is produced as a by-product, wherein the chilled air is introduced into a crystallization chamber. Also injected into the chamber is a spray cloud of seawater droplets, which has been pre-chilled by heat exchange with the cold chamber walls, and which is then circulated and exposed to the chilled air in the chamber. The sizes of the droplets can vary, but are preferably predetermined, along with the relative temperatures, flows and speeds of the spray and chilled air, such that when the droplets are circulated within the chilled air, and settle at the bottom of the chamber, they are deposited at slightly above the eutectic temperature. This way, the ice/snow mass that forms at the bottom of the chamber will consist of frozen ice crystals, and a residue of salt water brine, which can runoff from the mass, either from the sides, or through any voids or channels that may form within the mass

7. Desalination method and system using a Continuous Helical Slush Removal System

US 2010-0018247 A1

January 28, 2010

Australia 2014202087

April 14, 2014

South Africa 2009/01223

July 24, 2009

(Helical Screw)

8. Mineral recovery system for desalination

US 8,677,769 B2

March 25, 2014

(Mineral Recovery)

The invention relates to a mineral recovery system that processes concentrated brine leftover from a desalination system, wherein the brine is introduced into a crystallization chamber in the form of a spray, and wherein by mixing it with super chilled air from the desalination system, the droplets will flash freeze and form a super chilled slurry mixture, wherein an agitator for mixing the super chilled slurry is provided at or near the bottom of the crystallization chamber. Multiple stilling chambers are preferably provided for further separating the water from the concentrated brine, to produce a heavily concentrated sludge containing mineral solids that can be removed from the brine and recovered for commercial purposes

9. Method and Apparatus for Removing Carbon Dioxide from Coal Combustion Power Plants

US 2009/0205364 A1 (Pending) August 20, 2009

Australia 2009206700 January 23, 2009

South Africa 201005995 January 23, 2010

(CO₂ removal)

High mass flow of super chilled air is heat exchanged with the effluent from the smoke stack to extract the CO₂ as a mass of dry ice crystals. The dry ice is used in a variety of commercial industries.

10. Method and Apparatus for Sequestering CO₂ Gas and Releasing Natural Gas from Coal and Gas Shale Formations

U.S. 8,839,875 September 23, 2014

US 14/452,845 (pending) November 27, 2014

A system for sequestering CO₂ gas and releasing natural gas from underground coal and/or gas shale formations using CO₂ gas captured from the flue gas of a coal burning power plant, and processing it to produce cold liquid pressurized CO₂, and injecting the cold liquid CO₂ under pressure to create fractures within the formation and causing the CO₂ to be adsorbed into the coal or gas shale and CH₄ to be desorbed, released and recovered. No high volume of water, no toxic additives to the water and no sand proppants are used for hydro-fracture of the gas shale or coal bed.

11. Method and Apparatus for Using Solar Energy to Enhance the Operation of the Energy Storage System

US 8,024,928 B2 September 27, 2011

(Solar Enhancement)

Nearby solar thermal energy collectors circulate fluid to pressure vessels during daytime

12. Method and Apparatus for Using Compressed Air to Increase the Efficiency of a Fuel Driven Turbine Generator

US 8,833,083 September 16, 2014

Australia 2009302875 (pending) May 5, 2011

Chinese ZL200998049023.6

October 6, 2009

(CTT)

Chilled air from turboexpander combined with ambient air, centrifuged and then sent to electricity Generator Set (GenSet)

13. Wind Turbine Station and Tower with Vertical Storage Tanks

US 14/849,685 (pending)

September 10, 2015

(Vertical Tanks)

14. Method and Apparatus for Using Pressure Cycling and Cold Liquid CO₂ for Releasing Natural Gas from Coal and Shale Formations

US 8,833,474

September 16, 2014

US14/458,432 (pending)

December 4, 2014

(CO₂ pressure cycling)

Higher injection pressures than are conventionally used rubblelizes the stratum so that it has more than a hundredfold increase in permeability that permits more rapid removal of the natural gas. Furthermore, the horizontal injection of the LCO₂ jet avoids the expense of horizontal drilling. The emphasis herein is the higher pressure used in fracturing to induce rubble.

15. Method and Apparatus for Using Frozen Carbon Dioxide Blocks or Cylinders to Recover Oil from Abandoned Oil Wells

US 14/671,424 (pending)

September 17, 2015

(CO₂ ice blocks)

16. Method and Apparatus for Using Wind Energy or Solar Energy for an Underwater and/or for an Under Seabed Compressed Air Energy Storage System

US 13/610,044 (Pending)

September 11, 2011

(CAES System under water)

This disclosure refers to the CAES system that is off shore and wherein the pressure vessel and pipes are under water or that are under the soil bed below the water. This permits less costly reinforced concrete pressure vessels. Also the heat input is obtained from the warm seawater via the compressed air pipe between the seabed and the surface.

17. Method and Apparatus for Integrating On-Shore Green and Other On-Shore Power Sources with a Compressed Air Energy Storage System on a Floating Power Plant.

(Floating Power Plant)

The invention relates to a method and apparatus that shares the cost and areal footprint of the barge floatation platform for a floating power platform (FPP) with the cost and aerial footprint of the high pressure vessels of the compressed air energy storage (CAES)

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AERODYNAMIC

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WAVE MOTION IN FLUIDS

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WAVE MOTION IN GASES

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MICROWAVE GAGE

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Haber, J., Legg, M., Lieberman, P., Earthquake Risk Assessment, March-April 1989 Issue of Risk Management Report.

SHEAROGRAPHY

Lieberman, P. Shearographic Nondestructive Testing, 12th Aerospace Testing Seminar, March 1990, Institute of Environmental Sciences, Manhattan Beach, CA.

ELECTROMAGNETIC INTERFERENCE (EMI) AND ELECTROSTATIC DISCHARGE (ESD)

Lieberman P., Gray P. and White J., "Testing Air Bag Inflators for Your Car", Evaluation Engineering, August 1994, pp 92-97

Lieberman P., Electrostatic Discharge Testing, EMC Test and Design, February/March 1995, pp 22-24.

Tran D., Boller D., and Lieberman P., "Jet Engines Need Custom EMI Chambers", Test and Measurement World, February 15, 1997, pp 13-14.

NARRATIVES

Experience

2000 – 2019

EnisEnerGen, Enis WaterPure, Enis Gas

Water Purification
Compressed Air Energy Storage
Waterless Fracking

1999- Present

Lieberman Research Associates (LRA), President

\$75,000 Award from California Energy Commission, Energy Innovations Small Grant (EISG) Program, Public Interest Energy Research (PIER) Grant Number 53718A//03-24, "Test and Evaluation of Heat Transfer Parameters for the Transportable Compressed Air Energy Storage (T-CAES) System" November 15, 2004

Was funded by COGEMA Engineering Inc (D.O.E.) for **accelerated ageing tests** of EPDM Hose-In-Hose Transfer Line transferring toxic and radioactive waste at the Hanford Reactor Site.

Prepared several system studies to support patent for pneumatic storage of electrical power generated by wind turbines, solar thermal coils and photovoltaic cells.

Submitted Proposals:

- Storage System for Electrical Power Generated by Wind Turbines
- Wind Concentrator for Application of Wind Turbines to Low Speed Wind Sites
- Ultrasonic Lamb Waves for Non-Destructive Testing
 - Air Force \$100,000 SBIR
 - Navy \$1,400,000 BAA
- One-way Laser Energy Transmission through Aerosol Cloud for Concealment during Downed-Pilot Rescue.
- Remote neutralization of Land Mines by High Intensity (400-psid), Controlled Pressure History Airblast (>10-milliseconds) with High Directivity.
- Terrorist Bio-Toxic Aerosol Cloud Release, Its Tracking, Engulfment and Destruction.
- Acoustics and Vibration as Enhancement of Water Desalination.
- Helical Vortex Motion Applied to Incoming Saline Solution Around the Reverse Osmosis Membrane Enhances Desalination System.

Developed a theory and a test matrix for an acoustic whistle based on an oscillatory capillary wave jet. The device was tested and operated successfully (for SARA).

1981 - 1999 National Technical Systems (NTS), Chief Scientist

Dr. Lieberman was elected a member of the Fluid Mechanics Technical Committee of the AIAA, 1991-1996.

Member of the AIAA Hypersonic Technologies and Aerospace Planes (HYTASP) Subcommittee since 1991.

Dr. Lieberman chaired the session entitled, "Experimental Methods" for the 26th AIAA Fluid Dynamics Conference, June 1995, San Diego, California.

Dr. Lieberman organized and chaired a morning and evening session of the AIAA 31st Aerospace Sciences Meeting in Reno Nevada, January 1993, entitled, "Hypersonic Vehicle System Issues."

Dr. Lieberman chaired a morning session of the AIAA 33rd Aerospace Sciences Meeting in Reno Nevada, January 1995, entitled, "Wind Tunnel Measurement Systems"

Dr. Lieberman prepared and presented a six-day course, "Dedication of Components for Nuclear Power Plants", in Tainan, Taiwan during the summer of 1992.

Dr. Lieberman is Chief Scientist at NTS. He has participated in the complete range of Military Standard testing of components for vibration, shock, pyroshock, acoustic, temperature, temperature shock, temperature and humidity, temperature and altitude, sand, rain, immersion, salt water spray, and solar radiation environmental tests. He also performs reliability analyses of

hydraulic and electrical systems, accelerated aging test design, and failure modes and effects analyses. In addition he supports the EMI and electrostatic discharge test facilities. Most recently he has been selecting processes for extracting hydrocarbons, solvents and heavy metals contaminants from both the vadose and saturated soils for use by the NTS Environmental Services Group.

Dr. Lieberman evaluated nuclear reactor components rated for a 10-year lifetime, to check if they can be extended to 40-year lifetime. These activities included a Failure Mode and Effects Analysis (FMEA) and Fault Tree Analysis.

He designed and patented a circular progressive wave tube for testing the Titan IV combustion chamber and nozzle extension wherein an oscillating shock in a supersonic jet generates the high intensity acoustic and ultrasonic power. The acoustic power exceeds that available from conventional modulators, especially in the range of 8,000 to 80,000 Hz. When one oscillating supersonic shock generator (OSSG) is driven by a gaseous oxidizer and another OSSG is driven by a gaseous fuel, there are extremely high temperatures and high acoustic-combined environments which can be obtained in an uncontained space for large and small test articles, with easy access for instrumentation and observation.

He designed a progressive wave facility that generates 190 dB OASPL. A train of blast waves of controlled periods is the basis for its operation. The facility can attain 195 dB OASPL for more than 10 seconds.

Dr. Lieberman reviewed the vacuum extraction, Bio-venting sparging and composting techniques for eliminating hydrocarbon and explosives contamination plumes in the soil. The vacuum extraction technique was selected for the contaminated soil at the Hartwood, Virginia site. Bioremediation proposals were prepared for the USEPA and DOE wherein heavy metals and radionuclides were in the contamination plume. For the Michigan Air National Guard Site in Alpena, Michigan he prepared ozone, oxygen, moisture, bioremediation and thermal treatment combination to remove high molecular weight hydrocarbons from vadose soil and from TPH laden soil and from TPH and lead laden soil; and a lime stabilization and solidification treatment for the leaded soil.

He designed a nuclear air blast simulator to test large shipboard antennas. The shock front was generated by the rapid fracture of a panel of brittle glass, with initial matched pressures on either side of the glass panel. The stored volume of gas and vent hole size controlled the positive phase duration. Designs were prepared to simulate the positive and negative phases of the blast wave. The concept can also be used to test large glass panels for response to sonic booms.

Dr. Lieberman, with Ron Bocksruker, designed a burner for testing aircraft fire wall panels subjected to 550-psia stagnation, pressure and 3,000 °F flames. He also designed and built an ejector system to test the altitude performance of a special Marquardt cryogenic power plant component for NASP.

Dr. Lieberman conceived, wrote proposals, and was awarded such projects as holographic filters for eye protection against laser weapons, shearography for non-destructive testing; giant holographic mirrors that concentrate incoming optical signals at specific locations for target identification; holographic mirrors for solar cells and heating working fluids and camouflage coatings. He also developed an optical displacement gage. He also developed a microwave

displacement history gage for automotive crash tests, and a piezoresistive pressure gage (5,000 psid to 450,000 psid) using ammoniated water as the sensing element. He developed a fiber optic cable to detect gas leaks from gas pipes. He evaluated a hydronic heat pump system for saving air conditioning and heating costs in a test HVAC network. He also developed an approach to pyrotechnic shock-plate testing to deliver specific shock spectra.

Dr. Lieberman participated in several large scale studies: Shallow Tunnel Basing Mode for Peacekeeper, the Cofferdam Concept Test Program for an upgraded Minuteman, the Titan IV Flame Deflector Study at Test Stand 1-C, and the DNA Shock Isolation Study for ICBMs in Missile Silos under Advance Threats.

Dr. Lieberman developed the concept of the NTS explosive-driven shock tube system that delivered pre-cursor airblast waveform loads to 1/6-scale models of the Hard Mobile Launcher. The grouser designs of four aerospace companies were evaluated.

1969 - 1981 TRW Systems Group, Vulnerability and Hardness Laboratory

Dr. Lieberman held the position of Senior Program Manger at TRW. Dr. Lieberman partook in the Peacekeeper Program at BMO involving acoustic, seismic, vibration, and infrared measurements as part of the Position Location Uncertainty Program of the Mobile MX in Multiple Shelters.

Dr. Lieberman was the Technical Director of Project HAVE NAME. This project investigated the electrical hazards associated with the release of carbon/graphite fibers from aircraft composite structures. The instrumentation aspects required the application and fielding of infrared light emitter diode detector systems, laser detector systems, radar systems, microwave free-space and microwave line-guided detector systems, high voltage discharge detector systems, infrared imaging camera, and iridium tagging. The system aspects required development of fiber cloud environments in free space as well as in duct systems and compartments, selection of sensitive electrical equipment in these calculated environments, arrangement of this equipment into a fault tree, specification of probability of survival and confidence in this probability of survival for each component over a range of environments, and the Monte Carlo computer calculation of the overall system survivability. These calculations were performed for aircraft, missile facilities, control centers and buildings. Data for the systems studies were obtained by participation in a series of field and laboratory tests at Rome Air Development Center, Naval Surface Weapon Center/Dahlgren Laboratory, Capistrano Test Site, TRW and Naval Weapons Center/China Lake. Additional data was obtained by his participation in the HAVE NAME Aircraft Accident Investigation Team. As Technical Director of Project HAVE NAME, Dr. Paul Lieberman evaluated the effects of selected breakdowns in the Autovon System in Feldberg, Donnersberg, and Kaiserslautern, West Germany; in Uxbridge, England; at NORAD, and at SAC Headquarters. In addition to the system survivability study, he evaluated communication equipment component operations in the Project HAVE NAME environment for the communication equipment aboard the B-52, KC135 and F-4.

Dr. Lieberman performed a concept and feasibility study of giant shock tubes for application to the MX System. He developed a 50-foot diameter by 2,000-foot long 1,200-psi blast facility that was practical for full-scale tests of the MX missile for AFWL/CERF. Structural and gas dynamics, as well as cost considerations, were required. He also performed a concept feasibility

and design study and supervised the construction of a 1,000-psi blast facility, 9-feet diameter by 7-feet high, test volume, at Hill AFB for structural tests of the Minuteman splice cases. During a series of field tests in Colorado, Dr. Lieberman developed infrared and microwave instrumentation for measurement of the dust cloud concentration histories that were generated by 50-psi airblast waves from 100-ton explosions. He also introduced the concepts of mass addition and protuberances to attenuate airblast in the MX tunnel basing system, and verified the concept in a series of shock tube tests. He has performed analyses to obtain a closed form analytical solution of stress wave transmission in bilinear, layered media. He has also performed analyses to determine the strain-rate sensitivity of concrete to fast rise-time applied loads, the silo closure slab motion through frozen debris for gas pressurized driver system, the spall criteria in a concrete slab for fast rise-time high intensity thermomechanical waves, the response of the missile launch facility slab closure to impact by high velocity large rocks, and the criteria for placing two silo structures in the rock test bed with no coupling.

1957 - 1969 IIT Research Institute

Dr. Lieberman was a scientific advisor at IIT Research Institute where he conducted study and field test programs leading to the development of instrumentation and facilities in support of HEST and high explosive tests in soils and in rocks. He participated in and was a Project Engineer on several chemical and nuclear detonation test programs including MINE SHAFT, FLAT TOP II, FLAT TOP III, FLAT TOP I, TINY TOT, PILED RIVER and RED HOT at the Nevada Test Site.

1955 - 1957 Rocket Engine Test Laboratory

At Rocket Engine Test Laboratory, Edwards AFB, Lt. Lieberman established the liquid oxygen boil-off of the Atlas missile and missile test stands, and evaluated preventative measures; he evaluated testing techniques for a 100,000 lb thrust fluorine-ammonia rocket propulsion system, and designed a facility for duplication of missile free flight vibration and altitude pressure conditions.

1955 - 1956 Princeton University

At James Forrestal Research Laboratory, Princeton University, Paul performed analyses in determining the natural frequency of combustion instability associated with the shape of the subsonic portion of convergent-divergent nozzle configurations.

Summer 1954 Bell Aircraft Corporation

During the summer of 1954, Mr. Lieberman developed and used an electrical model of heat transfer in nucleate boiling burnouts associated with coolant fluids in tubular rocket motor combustion chambers.